AN OPEN PLATFORM FOR STRUCTURAL SIZING OF COMPOSITE AIRCRAFT WING.

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Content

Aircraft Conceptual Design

OCAMSS - Purpose

OCAMSS - Design

OCAMSS - Layout

OCAMSS - Modules

OCAMSS - Process Flow

Future Plan and Improvements

Aircraft Conceptual Design

Preliminary Sizing

Geometric Modeling

Aerodynamic Analysis

Weight Analysis

Propulsion Analysis

Flight Performance Analysis

Cost Analysis

Environmental Analysis

A process involving Multidisciplinary analysis and procedures, enabling implementation of Academic Knowledge.

OPTIMIZER DATABASE

PURPOSE

AN OPENSOURCE PLATFORM A MULTIDISCIPLINARY PROCESS



A platform for **Students** and **Researchers** to implement modular tools.

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A platform that can be **Integrated** with other Conceptual Design Suite and existing Code.

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Researchers to implement modular

A platform for **Students** and



A platform that can be **Integrated** with other Conceptual Design Suite and existing Code.



A platform that can be **Extended** to serve as a Conceptual Design Suite.

DESIGN A PYTHON BASED PLATFORM AN EXTENSIBLE PLATFORM SIMPLE TOOLS

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A set of Core Modules and extensible and optional Dynamic Modules

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A Domain specific Documentation.



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OCAMSS Layout



Python PyQt Numpy SALOME Pyevolve

DATA Managers

CAD

PARAMETRIC

A parametric CAD Builder to design the Wing from given parameters with Database of parameters as Airfoil Data etc.

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SALOME MODULE : geompy

geompy.SubShapeAll
geompy.MakeEdge
geompy.MakeWire
geompy.MakeFace
geompy.MakeSketcher
geompy.MakeSewing
geompy.PointCoordinates
geompy.ChangeOrientation

ALGORITHMS AND HYPOTHESES

A set of Algorithms for the computation of Meshes based on provided set of Hypotheses. API to create and modify them.

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The Group Management Tools are of real importance for applying Forces, Boundary Conditions , Physical Properties etc.

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GROUP MANAGEMENT

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Mesh Visualization Module* of SALOME provides a set of options for Mesh visualization.

SALOME MODULE : **smesh StdMeshersDC** Python package provides interface to the standard SALOME Algorithms.

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MED supports eight element shapes: point, line, triangle, quadrangle, tetrahedron, pyramid, hexahedron, polygon and polyhedron. Each element may have a different number of nodes, depending on whether linear or quadratic interpolation is used.

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Standard Elements and Algorithms supported and provided with SALOME are about sufficient concerning our Physical Models. MED supports eight element shapes: point, line, triangle, quadrangle, tetrahedron, pyramid, hexahedron, polygon and polyhedron. Each element may have a different number of nodes, depending on whether linear or quadratic interpolation is used.

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Monitors and Controls the complete Structural sizing Procedure and provides the main interface to the platform.



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DATA MANAGERS

Communicates with other modules through a set of Data Manager Modules and provides tools to configure the Data Manager modules.

- Responsible for generation of input and output files in required formats.
- All user input files in xml format, both Python and PyQt provides great tools to manage them.
- MED Data Model used for internal data exchange.
- MED File management is implemented using C API and MED Memory management is done with Python API

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DEFINE FRAMEWORK

Provides interface to design the framework and define the calculation scheme for the procedure.

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- Define Design specifications.
- Import Preliminary analysis Data.
- Define Load-cases and constraints.
- Define calculation scheme using xml file and execute scheme using SALOME YACS module Python API.
- Define Solver Config./ Command file in xml format.
- *Interface to modular Solver Design.

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SUB-MODULES

It has two sub modules, the Optimization module and the Guidance module.



OPTIMIZATION MODULE

- Carries out the sizing procedure depending upon the constraints and given base design.
- Incorporates an Improved Genetic Algorithm¹ where local improvement is inserted into Standard GA.
- Finite element calculations in local search are replaced by regression analysis.
- 'Local' defines individuals in two ways :
 - i. Individuals which are different from the best individual by **only one variable**.
 - ii. Individuals in which all the values of the variables are the same as those of the best individual, but in which two are interchanged.
- **Pyevolve** Genetic Algorithm framework can be modified to perform the optimization.





 C. C. Lin and Y. J. Lee, "Stacking sequence optimization of laminated composite structures using genetic algorithm with local improvements.," Composite Structures, no. 63, pp. 339-345, 2004.



GUIDANCE MODULE

- Guides the user during the complete procedure and reports necessary feedbacks or errors as required during the procedure.
- Monitors data at every step and provides the user with information to move forward in the process.
- Depends upon the Documentation of modules, Design specifications, constraints, data from previous analysis and statistical data to provide the feedback.
- Introduction of special Guidance objects using Python Dictionary Type into the framework for reference and comparison.

PRE-REQUISITES

As per the requirement of the framework set up the platform with pre-requisite scripts, modules, relevant guidance objects etc. from the Main Interface.

- Creation or Modification of Data Manager Modules .
- Solver Modification and Integration requirements.
- Refinement of CAD Builder or MESH Module.
- Availability of Constants and Statistical Data

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DESIGN SPECIFICATIONS

Define the Design Specifications and import the Preliminary Analysis Data (xml).

CAD BUILDER

Choose a Base Design to modify or create a completely new Design using the CAD Builder Module and the various Data and constants provided.

Airfoil Data	
1.0000000	0.0000000
0.9950000	-0.0036142
0.9900000	-0.0037896
0.9950000	0.0033617
1.0000000	0.0000000
Airfoil Geome	etric Equation

04

CAD BUILDER

Choose a Base Design to modify or create a completely new Design using the CAD Builder Module and the various Data and constants provided.

MESHING

Import the CAD Model, mesh the model using the appropriate algorithms based on proper hypotheses that can support the Physical model of your Framework.

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- Spar and Rib Caps : Rod Elements.
- Spar and Rib Webs : Panel Elements.
- Engine Mount : Bar Element (Extensional, Bending and Torsional Stiffness)
- Wing Covers : Polygon membrane Elements.

ANALYSIS

05

Define the constraints, loadcases and the calculation scheme. Create the Config./ command file for the Solvers. Start and monitor the analysis.

xml config./command file (representative)-

```
<material id="1" name="MAT01" >
<elas_mod type="double">2.06E11</elas_mod>
<p_ratio type="double">0.3</p_ratio>
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</material>
<mesh id=""> </mesh>
<model id ="" mesh=""> </model>
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Optimization Variables -

- Sandwich depth
- Thickness of layer t_0 , t_{phi} and t_{90} associated with angles 0, phi and 90.
- Stacking sequence and
- Orientation angles gamma and phi

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POST PROCESSING

Use the Analysis output and calculate the secondary variables. Import the data into SALOME Visualization Module to carry out the post processing.

Future Plan and Improvements

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Full Deployment and Validation of Concept.

A complete Database instead of collection individual data files.



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Academic references in Guidance Module

FE Tool for Static Analysis of Trusses



A simple application for Finite Element Static Analysis of Trusses with One Dimensional Rod Elements .

Python PyQt numpy







Thank You